Normal range	Comments
1200 to 1300°F	EGT is approximately 200°F lower than cruise when full throttle applied.
1350°F	under 200 HP
1550°F	high performance
70 to 90°F	fuel injected
120 to 150°F	carbureted
1600°F average	100° higher than EGT
350°F (OAT 60°F)	normally aspirated
410°F	Turbocharged
50 to 70°F	100° with gasket probes
200°F	oil cooler thermostat opens at 180°F
-40°/minute	tightly cowled
-55°/minute	Bonanza
-200°/minute	helicopter
of less than -60°/minute. You w will shift from front cylinders (ill find that the cyl during climb out) t
NOTE: If one CHT is reading 20° to 50° above or below the others, this	or below the o
	Parameter Normal range Comments EGT at Takooff 1200 to 1300°F EGT is approximately 200°F cover than cruise voven than the curvant cruise tabek cooling rate of less than -60°/minute waintain a cooling rate of less than -60°/minute. You will fluid that the curvant corum volume

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CHT probe is occupying the socket as the factory original probe. Chart your dealer.

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Engine Diagnosis Chart The following chart will help you diagnose engine problems in your aircraft (unit is in 'Percentage View' mode for these pictures).

							Uispiay	aucrari (unit is in 'Percentage View'
		sss of EGT for ue cylinder, <i>no</i> <i>yital EGT</i>	oss of EGT for ne cylinder. ngine rough	EGT Increase o lecrease after gnition system naintenance	75" to 100° EG rise for one cylinder during flight		Symptom	Percentage V
	lly; faulty	Failed probe or failed wire harness.	Stuck valve. Other cylinders are okay.				Probable Caus	mode for the
Check compression.	Have valve lifter or rocker arm checked.	Swap probes to determine if probe or wire harness is bad.	Have valve train checked.			Action	-	se pictures).
		Image: State of the state	Loss of EGT for Failed probe digital EGT ECT ECT Failed wire hamess. ECT or ne winder, no pr ECT alled wire hamess. ECT or one Copening fully; faulty Valve lifter. Paul A and the probe Copening fully; faulty Paul A and the probe Copening fully; faulty Paul A and the probe Copening fully; faulty Paul A and the probe Paul A and the probe Copening fully; faulty Paul A and the probe Paul A and the probe Paul A and the probe Copening fully; faulty Paul A and the probe Paul A and the	Loss of EGT for Engine rough Engine rough Engine rough Engine rough Engine rough Engine rough Engine rough Engine rough EGT for Failed probe electron or alled wire hamess. EGT Decrease in Decrease in Collinder no Permas EGT for one Collinder one Collinder at low Valve litter.	an an with with with with with with with with	an are send at a set of the tor one optiming plut or one optiming plut of the tor one optiming distribution. EGT increase after EGT or the tor one plut of the tor one optimitor, low EGT or the tor one optime rough plut of tor one optime rough plut of tor one optime rough plut of the term of the tor one optime rough plut of the term of the tor one optime rough plut of the term of the tor one optime rough plut of the term of term of term of the term of term o	Image: State Provided State Provid	Symptom Probable Cause Image: Symptom T5° to 100° EGT Spark plug not firing plug not firing plug to touling, faulty plug, with a symptom Image: Symptom EGT increase after plug not firing plug, with a symptom EGT increase after plug, with a symptom Image: Symptom EGT increase after plug, with a symptom EGT -> relating plug, with a symptom Image: Symptom EGT increase after plug, with a symptom EGT -> relating initian; high a symptom Image: Symptom EGT for system Stuck valve. Other inner, how EGT -> advanced ignition; low EGT for cough a symptom Image: Symptom Loss of EGT for system Stuck valve. Other inner, how or a system or alied probe glinder, no or an or alied probe glinder for one plug take valve not plug in take valve not plug for for one plug for for one plug for a now valve lifter. Balled with the namess. Image: Symptom Decrease in plug for for one plug for a now valve lifter. Low compression. Ch

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			renting and a second se			an San San San San San San			Display
		Sudden off scale rise for any or all cylinders	Rapid rise in EGT/CHT of one cylinder	High CHT on cylinders on one side of engine	Stow rise in EGT. Low CHT		Decrease in EGT for all cylinders	EGT and CHT not uniform	Symptom
or failed probe.	or Normalize view,	Pre-Ignition,	Detonation.	Obstruction under cowling.	Burned exhaust valve. CHT is low due to low power output.	Engine units set to Celsius	Decrease in ainflow into the induction system. Carb or	Dirty fuel injectors or fouled plugs.	Probable Cause
Check probe.	Change to Percentage view,	Full rich and reduce power,	Reduce power.	Check for improper Installed baffling, cowi flap misalignment or bird nests.	Have compression checked.	Check that the alarm limits are set to Celsius degrees	Check for change in manifold pressure.	Check injectors and plugs. Non-unitormity is normal for carbureted engines	Recommended Action

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Display (no picture)	<i>Symptom</i> Loss of peak EGT	Probable Cause Recommended Action Foor ignition or vapor Have magneto tested. In fuel injection system.	<i>Recommended</i> Action Have magneto teste
(no picture)	Decrease in peak or flat EGT response to leaning process	Detonation. Usually the result of 80 Octane fuel in 100 Octane engine.	Enrich mixture, Ireduce power and re- lean mixture. Repeat to find power setting where normal peak is obtained or run rich.
	Below 10,000 ft. full throttle causes EGTs to rise	Weak or defective mechanical fuel pump.	Apply booster pump. If EGTs drop, replace fuel pump.
	CHT more than Leaking ex 500°, EGT gasket blow normal. Adjacent CHT probe EGT may be low	haust ving on	Look for white powder around cylinder to determine leak area.
Torona Torona Channer Theadd	RPM RPM	arge DIF at low Blow by in cylinder APM fings	Check compression
Alarma			

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Alarms The EDM has programmable alarms. When a parameter falls outside normal limits, the display flashes its value and acronym. Once the parameter returns to normal limits, the flashing stops. If you've connected lights to the EDM's two enunciator outputs, they too will flash. The 'Remote FF Alarm' output (P4-11) activates when the fact remaining or endurance falls below your preset limit. The 'Remote Alarm' output (P1-12) activates with all other alarms. Note: Individual EGT's do not have separate alarms because they can assume different values depending on the installation and/or flight configuration—run up, "limb, cruise. However, there is an important alarm which occurs when a difference between hottest and coolest EGT's exceeds a programmed limit. This is called the 'DIF' alarm. This can quickly warn you when cylinder combustion becomes abnormal.

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When an alarm is displayed, *tapping* the STEP button will temporarily disable the alarm digital indication for the next ten minutes. When an alarm is displayed, *holding* the STEP button until the word OFF appears will disable that alarm digital indication for the remainder of the flight. See ' For Your Safe Flight

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Combustion that is too rapid leads to detonation and possibly pre- ignition. <i>Detonation</i> is abnormally rapid combustion where the file-air mixture explodes instead of burning uniformly. It causes the EGT to decrease and the CHT to increase, and can appear during the leaning process. It occurs under high compression from fuel with too low an octane rating, or from avgas contaminated by jet fuel. Fuel additives, such as lead, boost the octane rating and slow down the combustion process, producing an even pressure to the piston.	Highest priority CHT High CHT OIL High OIL temperature TIT High TIT OIL Low Old temperature CLD Excessive CHT cooling rate DIF Excessive EGT span VOLTS High battery voltage VOLTS Low fuel quantity remaining LO REM Low fuel endurance remaining Lowest priority FF Low fuel flow	Factory Limits and GPS' section on page 56 for a list of the alarms and their factory default settings. Alarm Priority If multiple alarns occur simultaneously, the higher priority alarm will temporarily "mask" the lower priority alarm(s). When an alarm occurs, note the cause of the alarm and tap the STEP button to clear the alarm indication so that you will be notified of any other alarm that might have occurred. The alarm priorities are as follows:

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Pre-ignition is caused by hot spots in the cylinder. Ignition occurs prior to the spark plug firing. The EDM depicts pre-ignition as a sudden red line of the EGT on the analog display. This may occur it one or more cylinders. The affected cylinder column(s) will flash while the digital display will show an EGT higher than 2000'F. At this temperature pre-ignition will desiroty your engine in less than a minute unless you take immediate corrective action.

Section 4 - Fuel Flow Option Operation

Fuel Management

Without a means of measuring accurate fuel flow, you must rely on the aircraft fuel gauges or total time of flight. Aircraft fuel gauges are notoriously inaccurate (they are only required by the FAA to read accurately when displaying *empty*). Determining fuel consumption by multiplying time of flight by estimated flow rate is, at best, an approximation, and assumes a constant fuel flow rate for each phase of flight. However, the EDM Fuel Flow Option uses a small, hurbine transducer that measures the fuel flowing into the engine. Higher fuel flow causes the transducer turbine to rotate faster which generates thousands of pulses per gallon of fuel, it can measure with high resolution the amount of fuel that flows into the engine. Prior to engine start you inform the EDM Fuel Flow Option of the known quantity of fuel onboard, it subsequently tracks all fuel delivered to the engine.

IMPORTANT: For EDM fuel calculations to be accurate, it is mandatory that you inform the EDM of the correct amount of usable fuel onboard the aircraft and confirm proper operation of the fuel flow transducer prior to and during flight. Do not rely on fuel flow instruments to determine fuel levels in tanks. Refer to original fuel flow instrumentation for primary fuel management information.

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Page 40 Engin		The three examples, shown below, depict different aircraft tank configurations and how you can update your EDM after refueling your aircraft. These are meant to be general guidelines. You are responsible for insuring that your usage of the <i>REFUEL</i> feature results in the EDM showing the correct amount of usable fuel remaining onboard the aircraft.	NOTE: If you forgot to update your EDM fuel remaining before starting the engine, you can still perform the REFUEL procedure. The EDM will automatically subtract any burned fuel from the REFUEL value you choose (not applicable to the 'ADD/SUBTRACT FUEL ONBOARD' feature).	Once you've selected your choice, tap STEP to accept it.	Choice 3) ADD/SUBTRACT FUEL ONBOARD?: This allows you to adjust your REMaining either up or down by holding or tapping LF. Use when adding a partial amount of fuel.	Choice 2) FUEL ONBOARD 86 GAL: This shortcut sets REMaining to the <u>sum</u> of MAIN and AUX (you preset AUX in Factory Limits setup).	Choice 1) FUEL ONBOARD 66 GAL : This shortcut sets REMaining to the MAIN tank value (you preset MAIN in Factory Limits setup).	On power-up, you will be prompted to enter any fuel you might have added to the aircraft (this process updates the REMaining and USED values). The EDM will flash <i>REFUEL</i> ? <i>N</i> . If you didn't add any fuel, simply tap STEP to exit, otherwise tap LF and pick one of the three choices below:	Start Up Fuel
Engine Data Management		t different aircraft tank rour EDM after refueling your idelines. You are responsible <i>IEL</i> feature results in the EDM sl remaining onboard the	DM fuel remaining before m the REFUEL procedure. any burned fuel from the able to the 'ADD/SUBTRACT	TEP to accept it.	VBOARD?: This allows you to x up or down by holding or g a partial amount of fuel.	UEL ONBOARD 86 GAL: This shortcut sets REMaining to the <u>sum</u> of MAIN and AUX (you preset AUX in Factory Limits setup).	UEL ONBOARD 66 GAL : This shortcut sets REMaining to the MAIN tank value (you preset MAIN in Factory Limits setup).	ter any fact you might have sthe REMaining and USED N. If you didn't add any fael, F and pick one of the three	



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Shock Cooling	Oil Pressure	Oil Temp	TIT, Turbine Inlet Temp 2	TIT, Turbine Inlet Temp 1	EGT, CHT	Total Fuel Used	Fuel Flow Rate	Time to Empty	Nautical Miles per Gal	Fuel Reserve at next GPS WPT or Destination	Fuel required to next GPS WPT or Destination	Fuel Remaining	RPM MAP	Difference between hottest and coldest EGT	Carburetor Temp	Intercooler efficiency	Compressor Discharge Temp	Induction Air Temp		Bus	Parameter
CLD -30 °/Min	OIL-P 52 PSI	01L-T 178 °F	2 TIT-L 1340 TIT-R 1340	1370 111	ест 1340 снт 376	USD 26.5 GAL	FUEL-F 13.5 GPH	EST. T TO E 02:45 H.M	ECON 13.0 MPG	EST. WP REM 77.3 GAL		EST. REM 37.2 GAL	RPM 2450 MAP 23.4	DIF 80 °F	CRB -22 °F	C-I 25 °F	CDT 300 °F	IAT 125 °F	OAT 81 °F	VOLTS 14.2	Example
Cyl. I.D. box indicates fastest cooling cylinder	Oil Pressure	Oil Temperature		Turbine Inlet Temp	EGT, left, CHT, right. Dot indicates cylinder	Since last refueling or trip total.	от КРН, ЦРН , РРН	Estimated Hours: Minutes Remaining at current fuel burn	Present with GPS interface and valid signal or MPK, MPL, MPP	interface Valid signal and way point	Present with GPS	in gallons, liters or pounds or kilograms		Dot indicates most widely deviating cylinder	Not available when CDT is installed	provided by the intercooler (if equipped)		Out of intercooler	°F or °C	Battery voltage	Comments

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Tap the LF button to select a choice, then Tap STEP button to execute. The EDM will show messages and percent complete indicators to keep you apprised of progress. When the download is complete the EDM may Page 47 For Your Safe Flight Page 47	 Downloading from Long Term Meethory From either Automatic Scanner or Manual Scanner mode, simply plug a properly formated USB Drive into the front panel jack (note: you must clear any alarns that are present for downloading to commence). There are three download choices: 'DUMP NEW': download all NEW flights since last download. 'DUMP ALL': a download ALL flights in the EDM memory. 'DUMP EXIT': EXIT the download mode. 	that may be set when you initially program your instrument. You may also program an <i>diveraft ID</i> that will appear in the output data file. We also program and <i>diveraft ID</i> to your aircraft registration number or recommend setting <i>Aircraft ID</i> is initially factory set to the EDM's serial number. You may change the recording interval from 2 to 500 seconds, number in flight (when you change the interval in flight, the current flight even in flight (when you change the interval in flight, the current flight file is closed and a new flight file is started at the new interval). During built-in tests, the EDM displays the Model, Date, Time, Percent Memory filled (since the last download), and the Aircraft ID.	For EDMs equipped with RPM sensors, recording begins when ECTs greater than800. For EDMs without RPM, recording begins when ECTs are greater than 850°F. The amount of data that the EDM can store will vary depending on how rapidly parameters change. The typical storage vary depending on which options are installed. When the memory can vary depending on which options are installed. When the memory becomes full, the oldest data will be discarded to make room for the newset. All data are time-stamped. The EDM contains a real-time clock	The EDM compresses and records all displayed parameters once every six seconds in Long Term Data Memory (note: you can change this rate to be 2 to 500 seconds). This data is retrievable by inserting a USB Drive to the jack on the front of the instrument and following the prompts. You can choose to retrieve 'ALL' the data stored in the EDM, or only the 'NEW' data recorded since your last retrieval. In either case, the selected data in the EDM is not erased. The data can later be viewed on EZTrends, a PC program available from JPI or over the internet.	Section 5 - Long Term Data Memory CAUTION: Do not insert anything but a USB Flash Drive Stick in the USB Port!	

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briefly show DONE. Wait a few seconds for the display to return to the normal engine monitoring mode and then remove your USB Drive. This completes the download.

Transferring from the USB Flash Drive to a PC

casy steps. To transfer your data from the USB flash drive to your PC, follow these

- 1. On your PC, start the EzTrends program.
- 3. In Ez
Trends, select the Move and Plot Data from Memory Stick 2. Plug in the USB flash drive into an available USB port.
- option.
- 5. Select the file you wish to plot and then select the flight in that file. 4. In the displayed list, find the USB flash drive and double click it.
- Refer to the EzTrends manual for details on how to use EzTrends.

Section 6 - Personalizing

Pilot Programming

To start the Filot Programming Procedure, simultaneously hold the STEP and LF buttons for five seconds. You will see the word PROGRAM for two seconds and then the sequence shown in the chart below. Tap the STEP button to advance to the next item in the list. Tap the LF button to select alternate values of that item. The shaded areas in the LF button to select alternate values of that item. The shaded areas in the example chart below pertain only to the Fuel Flow Option.

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KFACTOR 2		KFACTOR 1	MAP ADJUSTMENT 0.0	HP CONSTANT	OAT ADJUSTMENT 0	OAT DISPLAY	EGT DISPLAY		TRIP USED ?N	CAL TIT TIT +0	ORIG TIT ?N	AUTO SCAN	
KF2=29.90	KJ=7=29.90	KF1=29.90	MENT 0.0	T 108	OAT+i0 OAT-i0	л С С	N \$ ≺		N & Y	717 + 300 717 - 300	N Y V	e e	
Set using same method as K-FACTOK 1 above (for installations with fuel return transducer)	29.90 Adjust the flashing digit using LF button. Tap STEP to select next digit. Hold both buttons to save and exit KF-set mode.	Hold both STEP and LF buttons a few seconds to access KF-set mode (see below) or tap STEP to skip.	Hold STEP and LF for a few seconds until ADJUST shows. Adjust (by 43.0 in Hg) until MAP gauge equals the altimeter setting on a sea level airport. Tap Step to continue to the next step.	%HP display will change when HP constant is adjusted. Hold STEP and LF for a few seconds to access MAP calibration. Tap STEP to exit.	Adjust the indicated temperature up or down by up to 10°. For example, OAT+3 adjusts the OAT 3° higher.	Select degrees F/C. Tap STEP to accept or hold both STEP and LF buttons for a few seconds to access OAT calibration below.	Y-Yessets the digital display to one- degree resolution; N-No-sets 10°.	Y-Yes-Selects continuously accumulating total fuel USED, rather than resetting to 0 after EDM FILL process.	N—No—Selects resetting tuel costo w zero after performing EDM FILL process (after you filled aircraft)		Select Y if original ship TIT is being used. Tap STEP to accept. If Y chosen, then TIT is calibrated below.	Index rate (pause time) In the Automatic Mode. 0 disables the Automatic Mode. Tap 'STEP' to accept and go to next mode. Hold STEP and LF to access 'OR/G T/T ? N'.	

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Section	END? Y	ADJUST=50	DAYLCD ADJUST=50			TIME: 6 SEC	RECORD			01/01/15	DATE	-		18-23:59	Timo				TIME				GPS FORMAT		CARB = 2	- <u></u> .	CARB? N N	
7 - Progra	Y 🗢 N	-50 - +60	-90 - 100	2	ALPHA-	8 07 01 1	1-500			12/31/99	01/01/00-	_	20.00.00	33-59-59	00:00:00			430/530-WP	#6: GNS			430/530-WP	措: GNS		1+3		N⇔Y	
Section 7 - Programming Horsepower Constant	Y-Yes to exit, N-140 to lotton inc2		light conditions	Hold STEP and LF to save charges	tap "CHANGE" to change character/digit. Tap STEP to select next character/digit.	Hold STEP and LF to begin adjust. Hold or	Tap LF button to change data recuruling	save changes	select seconds. Hold STEP and LF to	Too LE to select minutes. Tap LF again to	Hold STEP and LF for 5 seconds until you	select seconds		see ADJUST. Tap LF to change nours.	Hold STEP and LF for 5 seconds until you	seconds to save format changes.	Hold both STEP and LF buttons a few	seconds to access Gro to maximum	Hold both STEP and LF buttons a rew	seconds to save format changes.	Hold both STEP and LF buttons a few	seconds to access GPS Forniat house	Hold both STEP and LF buttons a new	2 - highest Fuel Flow filter	1 - lowest Fuel Flow filter	yYescarbureted engine. Actus additional filtering to smooth out values.	NNoinjected engine.	

For EDM's equipped with FF, RPM, OAT and MAP, you will adjust the HP Constant once for your aircraft (NOTE: perform MAP adjustment prior to this process) Follow the steps below: Section / - Program

- Prior to takeoff, with the engine running, enter Pilot Program Mode (hold STEP and LF until 'PROGRAM' is displayed).
- 2. Tap STEP until you see HP CONSTANT 108 (your value may be different). Try changing the constant by tapping/holding the LF
- Page 50 Engine Data Management

ų, 4 Note: MAP function is an option and must be enabled in Factory Ν Perform calibration if the MAP on your manifold pressure gauge ω loesn't match the MAP shown on the EDM. Limits Mode. Section 8 - Programming Manifold Pressure (MAP) Ś 4 ÷ Q Once airborne, by reference to the Aircraft Flight Manual (AFM), <u>not</u> the EDM, setup a constant power setting of 70% in straight and level flight below 10,000 feet (be sure your mixture is Rich of Peak). Tap the STEP button to save and exit. constant. This adjustment will be performed while airborne. button. Note that the %HP gauge value changes as you change the to 70%. Note: the reading is the percent of maximum HP, not Adjust the HP constant until the EDM's %HP is as close as possible actual HP. If necessary, re-enter the HP constant mode (steps 1 & 2 above). Hold both the STEP and LF buttonsto change MAP adjustment. Tap STEP to index to MAP ADJUSTMENT +0.0 Enter the pilot program mode by simultaneously holding the STEP Do this on the ground with the engine turned off. and LF buttons for a few seconds. You will need to determine and set the absolute air pressure (this is A. Airport at sea level: adjust the MAP gauge digital display to the airport ATIS altimeter setting (or to your altimeter Kollsman window if no ATIS available) by tapping/holding the LF button. method 'A' or 'B' below: **B.** Airport <u>above</u> sea level: the table below shows MAP for a given field elevation (down the left side of the table) and altimeter what MAP displays). Depending on your airport elevation, use either The range is +/- 3 in HG. tapping/holding LF button. The range is +-3 in. HG. setting (along top row of the table). Find the closest entry in the Interpolate if necessary. Adjust the MAP gauge digital display by table to your field elevation and current altimeter setting. Alacting> 29.0 29.2 28.4 20.6 29.8 20.9 30.0 30.2 30.4 30.6 30.8 31.0 field dev. 29.0 28.2 28.4 29.6 29.8 28.0 28.2 28.4 28.5 28.7 27.0 27.1 27.3 27.5 27.7 8 29.9 30.0 30.2 7 28.8 28.9 28.1 7 27.8 27.9 28.1

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2 30.4 30.6 30.8 31.0 1 29.3 29.5 29.7 29.9 1 29.3 28.5 28.6 28.8 Page 51

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Page 52	6. Tap the STEP button to save. Section 9 - Programming use of Factory Original TIT probe and gaug should calibrate the EDM for that probe. The factory original TIT should calibrate the EDM for that probe. The factory original TI gauge insult calibrate the accuracy of the factory original gauge may be used the EDM rise concurrently. Due to the high input impodance of the EDM inso concurrently. Due to the high input impodance of the EDM inso concurrently. Due to the high input impodance of the EDM installed probe or gam romal cruise flight, record the difference between the factory installed probe or gam multiple on the accuracy of the factory installed probe or gam informat cruise hight, record the difference between the factory and UF put haven't already done so, start the pilot programming prosimilation start is water through the STEP and LF buttons for a few se you see the word PROGRAM for two seconds. Tap the LF button to seconds. Tap the LF button to seconds. Tap the LF button to seconds. AUTO SCAW AUTO SCAW AUTO SCAW Tap the LF button to lower the seconds. APP in advance ITT +0 ORIGIN ITT ?N ORIGIN ITT ?N ORIGIN IN Seconds. ORIGE to the LF button to lower the seconds.	seo 240 252 253 25 400 251 252 254 25 900 251 252 254 25 900 251 252 25 700 254 255 257 25 For this method, do not
Engine Data Management	(Kollsman window) setting struct that setting to the pressure. at sea level, and is not the same as your field elevation pressure. 5. Tap the STEP button to save. Section 9 - Programming use of Factory Original TIT probe should calibrate the EDM for that probe. The factory original TIT probe should calibrate the EDM and factory original gauge may be used concurrently. Due to the high input impedance of the EDM instrument, it original cruise flight, record the difference between the factory installed probe or gauge. In "romal cruise flight, record the difference between the factory installed probe or gauge. In "romal cruise flight, record the difference between the factory installed probe or gauge. In "romal cruise flight, record the difference between the factory installed probe or gauge. In "romal cruise flight, record the difference between the factory installed proceedure, by simultaneously holding the STEP and LF buttons for a few seconds until you see the word PROGRAM for two seconds. Tap STEP to advance sequence through the seconds. Tap STEP to advance of the LF button to seconds. Tap STEP to advance through the seconds. Tap STEP to advance of the STEP and LF buttons for a few seconds until you see the word PROGRAM for two seconds. Tap STEP to advance through the seconds to access the next sequence through the seconds to access the next sequence to book or the seconds. Tap STEP to advance of the STEP and LF button to lower the correction, hold the LF button to lower the correction, hold the UF button to lower the correction, hold the UF button to lower the correction, hold the UF button to east the procedure. ORG FTT 4 Tap ST	suon zad zad

Section 10 - Programming the Fuel Flow Option

Option is installed: **Fuel Flow Parameters** Three additional parameters may be set by the pilot when the Fuel Flow

fuel used will continue to accumulate for all subsequent flights. GPS Communications fuel data format. Accumulate-default is OFF: resets the fuel used to 0 every time you inform the EDM that the aircraft was refueled. With accumulate ON, K Factor-the fuel flow transducer calibration constant.

The K factor is shown on the fuel flow transducer as a four-digit number, which is the number of pulses generated per tenth gallon of fuel flow. Before installing the transducer, write down the K factor here

To enter the number into the EDM, place a decimal point two places from the right of the number. For example if the K factor ritten on the fuel flow transducer is '2912' enter 29.12 in the EDM K factor parameter field.

The K factor can be changed in the pilot programming procedure. When the K factor is changed during a trip, calculations of fuel used, fuel remaining and time to empty are not retroactively recalculated.

Fine Tuning the K Factor

account your aircraft's particular installation. Fuel hose diameters and The K factor shown on the fuel flow transducer does not take into different from that shown on the fuel flow transducer. lengths, elbows, fittings and routing can cause the true K factor to be

You must use the following procedure to fine tune the K factor.

1. Make at least three flights of about two to three hours each. Note the actual fuel used (as determined by topping the tanks) and the EDM calculation of the fuel consumed for each flight = USD.



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Accumulate Tetal—Trip Total Select TRIP USED? from the Pilot Program Mode. Select 'W' if you wish to display total fuel used since the last time you informed the EDM that the aircraft was refueled. Select 'V' to display total fuel used for an extended trip with multiple fuel stops. This selection affects only the USD parameter. Page 54 Engine Data Management	Every time you fine tune the K factor, record the measurements here: BEDM C Current K New K factor Filo's Date fiel used field used field of the factor filo's Date field used field used field of the	3. Record the current K factor here • the table below. 1. Calculate the New K Factor as follows: New K Factor = (<u>0 EDM fivel used</u>) x (<u>0 Current K factor</u>) (0 actual fuel used) New K Factor = (<u>0</u>) x (<u>0</u>)
t 'W' if you wish i the EDM that sets only the sment	nents here: factor Pilot's sectors. If you seconds until =29.90 mill the first digit xses): 29.00 00	and in

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GPS-C Comm settings select this from the Pilot Program Mode. The GPS-C setting selects the iormat of the *fuel data* output of the EDM. See page 62.

Section 11 - Programming Long Term Data Memory

If you haven't already done so, start the Pilot Programming Mode by holding the STEP and LF buttons for a few seconds until the word *PROGRAM* appears for two seconds. To change the date, time and user id for the Long Term Data Memory, tap the STEP button until the display shows *TIME*. Then set the data memory parameters as show below:

END? Y	<u>N</u>	TIME	HOUK:	YEAR:	DAY:	DATE:	TIME:	STEP
	N 123456	_		66 00	131	01/01/00 12/31/99	0 - 23:59:59	5
	Current Annuant of the first both STEP and LF buttons until the first character, character flashes. LF selects the first character, character flashes. LF selects the first character, strep movies to the next character. To Save, hold both STEP and LF for 5 sec.	himmet ID To change Alrcraft ID, hold	Record time interval, in seconds	24 hour time. We suggest you use Zulu time	Day (14) concepts 2000 through 2099)			Comments

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Section 12 - Factory Limits and GPS

Jactory Set Default Limits—Non-Primary JPI conservatively sets the default alarm limits below Lycoming and Continental recommendations. This is an example of default limits. Your actual aircraft alarm limits may differ from these:

1 temporanily	V CTED Littor will formorari		
	(Tr/a)	10 gal. kg. ltr. lbs	
		1111 044	
00.20 11.10	(n/a)	A5 min	
	9111120	(n/a)	MAP
MAP 46.3	20 in ha		BA1, 12 V
VULIO 11.0	16V	Vet	
10 TO 17 0	VZC	24	BAT 24 V
VOLTS 22.4	21		
DIL 070	500°F 280°C 1		
n non	Ļ	(n/a)	
CLD 00	enerimin -33°C/min.		
		(n/a)	
777 7720	+	90-r Joc C	
OIL 280	230°E* 110°C	-Ŀ	
100	450°F" 230 C	(n/a)	24
ARE CHT2	Course Dougo	Delant LOW LINI	Parameter
Alarm Example	Default High Limit		

When an alarm is displayed, tapping the STEP button will temporarily lelete that parameter from the sequence for the next ten minutes. When an alarm is displayed, holding the STEP button until the word OFF appears will delete that parameter from the sequence for the remainder of the flight.

Changing the Alarm Limits/Tank Capacity You may prefer to set your own alarm limits or set up your usable fuel in You may prefer to set your own alarm limits or set up your usable fuel in the main and, optionally, aux tanks. Follow the procedure outlined below to change any of the factory default settings. To start the alarm limit to change any of the factory default settings. To start the alarm limit procedure, after power up, wait until the EDM completes its self-test and procedure, after power up, wait until the EDM completes its self-test and is in the Automatic or Manual mode. If in doubt, tap the STEP button a for the transmission of the settings. few times. Then follow the steps depicted below:



The display will then sequence similar to the example chart below. Tap the STEP button to advance to the next item in the list. Tap the LF button to select values for that item. Hold the LF button to increase a

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numerical value; *tap* the LF button to decrease a numerical value. The haded areas in the chart below apply to EDM's equipped with the Fuel rlow Option. Example sequence and procedures for changing the alarm limits or main tank capacity shown below:

Ξ	nits or main tank	limits or main tank capacity snown below:	
	Tup STEP for	LF sequences through these value fanges	Description
	₂ ≺ 	N⇔Y	Restore factory defaults?
	EDM TYPE:	EDM730 ⇔ EDM830	Selects EDM730 or EDM830 display layout
_	RPM SENSOR	Y⇔N	RPM option, Y=enable
	MAP SENSOR	Y⇔N	MAP option. Y≃enable
		°F⇔°C	all engine temps. You
	IEMPS IN T		must also change the alarm limits to "F or °C.
	CHANGE		Enters a sub-menu to set limits. See table below.
	AVERAGE CRUISE RPM	75% OF RPM REDLINE ↔ 100% OF RPM REDLINE	Average cruise KEIN value used to adjust TACH TIME, When RPM
			faster than real time.
			Cruise RPM, TACH TIME
	-		real time
	LAST FLIGHT	7⇔999999	Once set, this value can never be set to a lower
	į		flight number to ensure that every flight has a unique number.
	FUEL UNITS	GAL⇒	Selects the units in all narameters where fuel
	GAL	LTR⇒	quantity or fuel rate is displayed
	MAIN TANK	666 0	Main tank capacity, in units selected
	AUX? N	N⇔Y	yyesaircraft has auxiliary tanks
	AUX TANK	999	Auxillary tank capacity
	R-FACTOR CYL=6	1-99	Set the number of cylinders. See page 61 for exceptions.

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|                        | Y 🗢 N                                                                                   | N⇔ Y                    | INDER 0 - 60 SECONDS<br>EOUT 5<br>DNDS           | FUEL USED 0 – 500 Displa<br>REMINDER 0 LTRS.                                  | LOP DFLT? N & Y Set to<br>of Pea                                              | REMEMBER     N ⇔ Y     Remember       CHANNEL     Is power       EXCLUSIONS     "Exclusion       topic.     topic. | CUSTOMIZE NEXT OR YES enters<br>DISPLAY? mode.<br>Display             | 3INE 1.0 99.9<br>VSTANT 1.0 99.9 | ENGINE HP 20 999 Set the | $CYL 7, 8, 9? N \qquad N \Leftrightarrow Y \qquad \qquad N - 4 or \\ Y - Allo$ |
|------------------------|-----------------------------------------------------------------------------------------|-------------------------|--------------------------------------------------|-------------------------------------------------------------------------------|-------------------------------------------------------------------------------|--------------------------------------------------------------------------------------------------------------------|-----------------------------------------------------------------------|----------------------------------|--------------------------|--------------------------------------------------------------------------------|
| Y-Yes to exit: N-No to | <ul> <li>N—no serial output:<br/>Y—Output real-time data<br/>to serial port.</li> </ul> | COGuardian sensor input | 0 seconds – no timeout,<br>require a STEP press. | Display a message each<br>time the number of GAL,<br>LTRS, etc has been used. | Set to Y if you want Lean<br>of Peak to be the default<br>LeanFind operation. | Kennemoers exclused<br>channels next time EDM<br>Is powered up. See<br>"Exclude parameters"<br>topic.              | nters customization<br>mode. See Customize<br>Display Sub-menu below. | Set the Engine Constant          | Set the engine HP        | N – 4 or 6 cylinders<br>Y – Allow up to 9 cylinders                            |

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st The alarm limits may differ from those shown here, depending on your type of aircraft.

## Advisory Limits Sub-menu

| Indementa,                       |             | COOLING -60 |
|----------------------------------|-------------|-------------|
| Cooling inner set in a more      | -5200 "/MIN | SHOCK       |
| Inclaiments.                     |             | TEMP 450°   |
| CHI mgn llinit, set il o         | 90 500°     | HIGH CHT    |
| IU IICIEIIEIIa.                  |             | DIFF 500    |
| EG1 UITEIEING IIsily as a        | 30 990"     | HIGH EGT    |
| FOT difference limit set in      |             | 12.0 VOLTS  |
|                                  |             | BATTERY     |
| Dattery low voltage initia       | 10.0 35.0   | LOW         |
| n-the lost voltage limit         |             | 16.0 VOLTS  |
| Set III 0.2 YOK II SI CITIZITIZI |             | BATTERY     |
| Battery nigh voltage white       | 10.0 35.0   | HIGH        |

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| DONE?                | LOW TIME TO<br>EMPTY 45<br>MINUTES              | LOW FUEL<br>REMAINING<br>10 GAL                                     | HIGH CDT<br>TEMP 300° | HIGH MAP 40   | AVERAGE<br>CRUISE RPM<br>2400   | HIGH RPM<br>2700   | LOW OIL-P             | LOW OIL              | HIGH OIL<br>TEMP 230°  | HIGH TIT<br>TEMP 1650°                                     |
|----------------------|-------------------------------------------------|---------------------------------------------------------------------|-----------------------|---------------|---------------------------------|--------------------|-----------------------|----------------------|------------------------|------------------------------------------------------------|
| NO / YES             | 0 60                                            | 0 200                                                               | 40 - 500°             | 25 90         | 75% - 100% OF HIGH RPM<br>LIMIT | 1000 3500          | 0 99 PSI              | 10 250°              | 40 500°                | 650 2000 °                                                 |
| FACTORY prompt menu. | Alarm limit in minutes for<br>low time in tanks | Alarm limit for low fuel<br>quantity in tanks, in units<br>selected | Increments            | when equipped | calculation                     | equipped with RPM) | in 1 pound increments | set in 5° increments | set in 5° increments.* | Also sets the maximum scale of the EGT and TIT bar graph.* |

## **Customize Display Sub-menu**

| FACTORY prompt menu.        |                 | DONES      |
|-----------------------------|-----------------|------------|
| Tap YES to return to        | NO / YES        |            |
| order.                      |                 | GAUGES     |
|                             |                 | TO FACTORY |
| Advinal IBI factory default | NO/YES          | RESET ALL  |
| Dectores linear nations to  |                 | GAUGES?    |
| by optione)                 |                 | LINEAR     |
| proferences (choices varv   | NEXT OR TES     | CHANGE     |
| V - Select linear dauge     |                 | GADGES     |
|                             |                 |            |
| Dargraph with set of a      |                 | LINFAR     |
|                             | 5 🗢 10          | MAXIMUM    |
| Have for widor EGT/CHT      |                 | POSITION=  |
|                             |                 |            |
| As described.               | TOP & BOTTOM OR | dγV        |
|                             |                 |            |

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# MAP, Fuel Flow Alarm Limits, Units, Fuel Capacity

MAP Overboost Alarm

inter the redline for overboost on turbocharged engines.

Selects the units in *all* measurements where fuel quantity or fuel rate is displayed. If you change this parameter, it does *not* change the numerical value of the fuel tank capacity. You must do this manually. For example if you change from Gal. to Lbs., the tank capacity will be interpreted as 50 Lbs. rather than 50 gallons; the EDM will not automatically convert 50 Gal to equivalent pounds. Fuel Flow Units (shaded area above)

## Main Tank Capacity

Enter the total usable fuel capacity of the main tanks in the fuel flow units selected.

answer "Yes," you will be asked to input the capacity of the auxiliary anks in the fuel flow units selected. Another scenario is if you have tank abs and sometimes fill only to the tabs. In this case you may set the auxiliary tank capacity to the difference between full tank capacity and If you do not have auxiliary tanks or tank tabs, answer "No." If you aircraft. the main and auxiliary tanks; it tracks only total usable fuel in the tab capacity. Note: The EDM does not differentiate fuel flow between

Select the value of the time remaining, in minutes, that triggers the fuel remaining by the current fuel flow rate. alarm. Time remaining is determined by dividing the current calculated Time to Empty Alarm Limit

## Low Fuel Alarm Limit

triggers the alarm. Fuel remaining is a calculated value and is dependent on what you've programmed into the EDM when refueling and the performance of the fuel flow transducer. Select the value of the fuel remaining, in the selected fuel flow units, that

## Carburetor?

slower response time to reduce sudden fluctuations in readings. The higher the number, the more filtering (the reading is smoother). carbureted or fuel injected. The filter for a carbureted engine has a Different response filters are used depending on whether your engine is

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Number of Cylinders This affects only RPM capable EDMs. Set RPM SENSOR PULSES PER REVOLUTION = 4 or 6 depending on your engine. Exceptions:

4 cylinder engine with dual (all-in-one) magnetos set to 8. 6 cylinder Lasar® ignition set to 12. 4 cylinder Lasar® ignition set to 8.

**HP and EC Constant settings** These adjustments set the parameters for the HP calculations in the EDM. Set the Rated HP for your particular aircraft (for example, a 300 horsepower engine: HP=300). Set the Engine Constant for normally aspirated or turbo normalized: EC=14.90 or for turbo boosted: EC=13.76. To change the EC field value, hold STEP and LF until the digit flashes. The flashing digit can now be edited using the LF button. Tap STEP to advance to the next digit. Hold STEP and LF until the change mode is exited, which will save the value.

**Entering the Tank Capacity** To initialize or change the tank capacities, hold the STEP button while turning on the power to the EDM. (Note: these can also be accessed via Pilot Program Mode). The following program steps will be displayed:

| Tap STEP | Tap the LF button to sequence<br>through these values | Comments                                 |
|----------|-------------------------------------------------------|------------------------------------------|
| MAIN     | 666 0                                                 | Main tank capacity, in<br>units selected |
| TANK     |                                                       |                                          |
| SIZE=50  |                                                       |                                          |
| AIVS M   |                                                       | Y-Yes-aircraft has                       |
| NOV / N  | 2                                                     | auxiliary tanks                          |
| AUX      | 0 250                                                 | Auxiliary tank capacity                  |
| TANK     |                                                       |                                          |
| SIZE=0   |                                                       |                                          |

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## **Navigation GPS Data Formats**

GPS radios communicate with other equipment in various formats to nite intended usage. You will need to determine what GPS mode is best suited for your installation and set up both your GPS and EDM for compatibility.

For incoming GPS data, the EDM automatically configures itself for one of three industry standard data formats, hence there is no EDM selection necessary for the EDM to receive the incoming data. The table below is for reference only:

|        |                                         |          |                                                                 |                                                     | _                                                 | _                                                      |                             |             | _                                              | _                                           | -           |     | •       |
|--------|-----------------------------------------|----------|-----------------------------------------------------------------|-----------------------------------------------------|---------------------------------------------------|--------------------------------------------------------|-----------------------------|-------------|------------------------------------------------|---------------------------------------------|-------------|-----|---------|
| binary | Northstar                               |          |                                                                 | Eormat                                              | Aviation Data                                     | 1                                                      | Data Format)                | (Marine Nav | NMEA-183                                       | rornut                                      | 1           |     |         |
|        | 1,200                                   |          |                                                                 |                                                     | 006,8                                             |                                                        |                             |             | 4,800                                          | DITIN                                       | Pand rate   | Lat |         |
|        | M1 setup select TNC EXTENDED, Juin Cite | MAV ONLY | terminator may be either <ck><lr> or <un> alone.</un></lr></ck> | C, D, E, I and L fillst dilatation interview of the | Uutput sentence type 1 toyour 1 in hyte. Sentence | "A Lost approach time 1" Remuired sentences are: A, B, | have sentences rund a rund. |             | Inis is the formation more RMA & RMB, GPS must | The second for most handheld GPS receivers. | Description |     | NTO EDM |

Note: If you connect a handheld, set its to output 'NMEA-183'.

The EDM sends formatted data to the GPS relating to fuel flow. This format must be selected in Pilot Program Mode to match what the GPS expects to receive. The table below depicts the available EDM choices:

# GPS-C Fuel Flow Format for GPS Bi-directional Comm

| GF 3-C F 0 |                                                 |
|------------|-------------------------------------------------|
|            | Data OUT OF EDM                                 |
| GPS-C      | Description                                     |
| 0          | No fuel data output                             |
| -          | Garmin (Shadin Miniflow tormat)                 |
| 2          | Alled Signal (format B) 10 waypoint only        |
| 3          | Amav/El Signal Format                           |
| 4          | Allied Signal (format C)                        |
| 5          |                                                 |
| 6          | 1-                                              |
| 7          | Gamin 430/530 GNA-80/GA-80/ 10 Destination only |
| 8          | Alled Signal (tormat b) 10 Desuitance on )      |

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Diagnostic Messages for Fuel Flow and GPS

The following tables depict various messages you may see when a malfunction or signal problem exists:

| : H:M                                | 0.0 FF                                           | 0.0 GPH                                                 | Fuel Flow Trai           |
|--------------------------------------|--------------------------------------------------|---------------------------------------------------------|--------------------------|
| Dashes indicate no fuel flow signal. | FF' flashing indicates loss of fuel flow signal. | Zeros indicate fucl flow signal is too low to register. | w Transducer Diagnostics |

## 

|                                                                                  | missing.                                     |
|----------------------------------------------------------------------------------|----------------------------------------------|
| No waypoints are programmed into the aircraft GPS receiver.                      | NO - WPT message and parameters REQ, RES are |
|                                                                                  | are missing.                                 |
| GPS receiver has insufficient signal for<br>valid data                           | NO - SIG message and                         |
| Verify correct output format setup in GPS receiver; check GPS connections.       | are missing.                                 |
| Communications are received by EDM<br>and the Auto-Protocol setup is in process. | NO - COM message and                         |
| GPS is off.                                                                      | are all missing from the scan.               |
| No communications from GPS receiver to                                           | Parameters REQ, RES, MPG                     |
|                                                                                  | GPS Interface Diagnostics                    |

# Navigation Data Ports for GPS Communication

# Navigation Data input to the EDM (an output from the GPS) EDM can accept RS-232, TTL, RS-423, RS-422 SDA.

Serial data format: 8 bit data, 1 start bit, no parity. Baud rates: 1,200, 4,800, or 9,600 depending on the GPS data output format. The EDM automatically detects the GPS data output format (independent of the EDM's GPS-C setting).

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Fuel Data input to GPS (an owtput from the EDM) RS-232 Serial data format: 8 bit data, 1 start bit, no parity. Baud rate: 9,600.

EDM Output format is normally fixed by the GPS-C number you chose, but may be over-ridden when the EDM auto-senses certain incoming GPS navigation formats. For example, if the EDM senses Northstar or NMEA-183 navigation data formats coming in, the EDM will not send fuel data to the GPS.

# Section 13 - Options Connector Pin Assignments

|                    |                 |                   |       | Engine                          |                              |             |
|--------------------|-----------------|-------------------|-------|---------------------------------|------------------------------|-------------|
|                    |                 |                   | _     | Ground at                       | blk 25                       |             |
|                    |                 |                   |       | data port                       |                              |             |
|                    | 11              | grn 9 MAP sig- /4 | gn    | <b>RS-232</b>                   | wht 24                       |             |
|                    |                 |                   |       | 14/28 vdc                       |                              |             |
| SWICH FF           | ç               | wht 8 MAP sig+ /2 | цw    | red 13 + Power                  | red 13                       |             |
|                    |                 |                   |       | alarm                           |                              |             |
| 8 SWITCH EGI       |                 | blk 7 Oil P term  | Þ     | gry 12 Remote                   | gry 12                       |             |
| Swilch COM         |                 | t 6 Oil P term    | wht 6 | TIT2                            | red 19 TIT2                  | yel 18      |
|                    | 4               | <u> </u>          |       |                                 | red 17 TIT                   | yei 16      |
| NU & FE return     | 214             |                   |       |                                 |                              |             |
| FF power           | red 5           | 14 MAP pwr /3     | red 4 | OAT                             | red 15 OAT                   | yel 14      |
| FF signal          | whit 4          | 13 RPM pwr /3     | red 3 | red 6 CRB/CDT                   | red 6                        | yel 5       |
|                    |                 |                   |       | IAT                             | red 4 IAT                    | yel 3       |
| RS-332 in          | 3               |                   |       |                                 |                              | .<br> _     |
| RS-232 out         | _               | עכ                | grn 1 | red 2 Oil Temp                  | red 2                        | yel 1       |
|                    | 110,            | sensor pin        | no.   | function                        | no.                          | <b>n</b> 0. |
| Function           | Pin             | t Function/       | Pin   | Probe or                        | Pin                          | Pin         |
| 1                  | !               |                   |       | ls -                            | cvlinder instruments         | cvlinder    |
|                    | _               |                   |       | installation manual for 7, 8, 9 | on manual                    | installatio |
|                    | ionanuoo uid-ci | 9-pin connector   | 9-pin | See                             | 6 cylinder engines. See      | 6 cylinde   |
| P4 Fuel Flow Opnon | P4 Fuel I       | P3 MAP-RPM        | P3 M  | or for 4 or                     | P1 25-pin connector for 4 or | P1 25-pir   |
|                    |                 |                   | ļ     |                                 |                              |             |

# Interface connections to selected GPS models

|   | TURNING CONTRACTOR of the |                      |                           |
|---|---------------------------|----------------------|---------------------------|
|   | FDM                       | Fuel flow conn Pin 1 | Fuel flow conn Pitt 2     |
|   | Amay 5000                 | Pin 4                | Pin 5                     |
|   | Camin 105                 | (nc)                 | Pin 4                     |
|   | Odmin 172                 |                      | n:- sc                    |
|   | Garmin 430 / 530          | Pin 57               | oc uld                    |
| 1 | Northeter M3D             | (nc)                 | Pin 6 (leave pin 11 open) |
|   | LOTAT IMPETITION          |                      |                           |
|   | TIPS GX50 / 60            | Pin 4                |                           |
|   | 010 01001 00              |                      |                           |
|   |                           |                      |                           |

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You can customize the linear gauges to your preference. Enter Program Mode, then tap STEP until *'END?* Y appears. Hold STEP and LF to enter *'FAC LM* mode. Tap STEP until you see *'CUSTOMIZE DISPLAY?'*. Tap Y to enter custom display sub-menu. Tap STEP until 'CHANGE LINEAR GAUGES' appears. Tap Y button. '*1=GPH*' appears (*GPH* may differ on your unit). Tap LF to choose a new function for this position. Tap STEP to accept and move to next position. Repeat until you are satisfied with the choices. Hold STEP and LF to save and exit. Now review your custom layout.

Examples of 'Position Numbers' for different layouts:





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| Page 66                | 800<br>711<br>WY                                                                                            | JPI offers both e-mail<br>support. Have your mod<br>JPI for a return authon<br>equipment.<br>J.<br>318<br>Co                                                                                                                                                                                                                       | <ul> <li>FAA Accident Prevention Program, FAA-P.</li> <li>Editors of Light Plane Maintenance Maga<br/>Belvoir Publications Inc., Greenwich, CT 0</li> <li>Lycoming Flyer Issue 53 dated January 93</li> <li>Technical Reviews and Original References<br/>For those pilots who have engineering backgroot<br/>oelow present the original research on the com<br/>represent the source documents for those with<br/>A. Hundere, "Autogas for Avgas," AOPA</li> <li>A. Hundere and J. Bert, "Pre-ignition and<br/>Aircraft Engines," SAE Quarterly Transac<br/>547-562, October 1948.</li> </ul>                                                                                                                                                                                                                      | Section 15 - Reference Reading<br>You may wish to know more about the e<br>EGT and CHT. The reading list below p<br>well as original references on topics that<br>General Overview<br>These references are readily available to<br>source of general technical information.<br>Tolatone Continental Motors. Envi                                                                                                                                                     |
|------------------------|-------------------------------------------------------------------------------------------------------------|------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|---------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| Engine Data Management | 800 345-4574<br>714 557 3805<br>Fax 714-557-9840<br><u>www.jpiustnuments.com</u> and <u>www.jpitech.com</u> | <ul> <li>JPI offers both c-mail (<u>support@jpitech.com</u>) and telephone technical support. Have your model and serial number ready when you call. Call JPI for a return authorization number RMA before returning any equipment.</li> <li>J.P. INSTRUMENTS Inc.</li> <li>3185 B Airway</li> <li>Costa Mesa, CA 92626</li> </ul> | <ul> <li>FAA Accident Prevention Program, FAA-P-8740-13.</li> <li>Editors of Light Plane Maintenance Magazine, EGT Systems,<br/>Belvoir Publications Inc., Greenwich, CT 06836. 1989.</li> <li>Lycoming Flyer Issue 53 dated January 93.</li> <li>Technical Reviews and Original References</li> <li>Torothose pilots who have engineering backgrounds, the references listed<br/>oelow present the original research on the combustion process and<br/>represent the source documents for those with technical interests.</li> <li>A. Hundere, "Autogas for Avgas," <i>AOPA Pilot</i>, Octobert, 1969.</li> <li>A. Hundere, "Autogas for Avgas," <i>AOPA Pilot</i>, Octobert, 1969.</li> <li>A. Hundere, "Autogas for Avgas," <i>AOPA Pilot</i>, Octobert, 1969.</li> <li>Scrion 16 - Technical Support</li> </ul> | Section 15 - Reference Reading<br>You may wish to know more about the effect of engine operations on<br>EGT and CHT. The reading list below provides general overviews as<br>well as original references on topics that may be of interest.<br>General Overview<br>These references are readily available to pilots and provide a readable<br>source of general technical information.<br>Teledone Continental Motors. <i>Envine Operation for Pilots</i> , from the |
|                        |                                                                                                             |                                                                                                                                                                                                                                                                                                                                    |                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     |                                                                                                                                                                                                                                                                                                                                                                                                                                                                      |

| $\frown$             | $\sim$                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      |                               |
|----------------------|-----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|-------------------------------|
| For Your Safe Flight | <ul> <li>J.P. Instruments Inc. (JPI) warrants all parts in your new EDM to be free from defects in material and workmanship under normal use. Our obligation under this warranty is limited to repair or exchange of any defective part of this unit if the part is returned, shipping prepaid, within three years for electronics and one year for probes from the date of orginal purchase. Installation labor is the responsibility of the aircraft owner. Homebuilt aircraft warranty starts when the aircraft or the varranty period.</li> <li>Under this warranty, JPI is not responsible for any service charges, including removal, installation, nor any other consequential darages. JPI incurs no obligation under this warranty unless a Warranty Registration Certificate describing the warranty nucless a Warranty Registration Certificate describing the warranty nucles a Warranty Registration Certificate describing the warranty nucles a Warranty Registration conficant on which, in the opinion of JPI, has altered or repaired the product in any way that effects the reliability or detracts from the serial number has been altered, defaced, effaced or destroyed.</li> <li>This warranty is in lieu of all other warrantics expressed or implied and other obligations of liability on JPI's part, and it neither assumes nor autorizes any other person to assume for JPI any other liability in connection with the sale of JPI products.</li> <li>To initiate the warranty the aircraft owner must submit a completed Data Logging Worksheet to JPI. Upon receiving a completed worksheet, JPI will initiate the warranty form the date of original purchase. Any yreplacement parts carry a warranty. For homebuilt aircraft the warranty starts when the aircraft is certificated for flight and noted on the warranty starts</li> </ul> | Section 17 - Limited Warranty |
| Page 67              | rts in your new EDM to be irec<br>protection or repair or exchange of any<br>to repair or exchange of any<br>probes from the date of<br>responsibility of the aircraft<br>when the aircraft is certified<br>anty for the balance of the<br>le for any service charges,<br>ther consequential damages. JPI<br>anless a Warranty Registration<br>uct has been subject to misuse,<br>damage in transit, handling or<br>t, has altered or repaired the<br>filty or detracts from the<br>ret whereon the serial number<br>royed.<br>In the sepressed or implied and<br>t, and it neither assumes nor<br>r JPI any other liability in<br>ter must submit a completed<br>sectiving a completed worksheet,<br>te of original purchase. Any<br>extends for the balance of the<br>ebuilt aircraft the warranty starts<br>tand noted on the warranty starts                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                            | <br> <br> <br>                |

| Page 68 Engine Data Management | <ul> <li>a. If you <u>do have</u> the fuel flow option, install the supplied P4 harness (JPI PN 700709-4) to the 'Slave' EDM. Install the <i>other</i> P4 harness (JPI PN 700709 no dash number) to the 'Master' EDM. Note: EDM &lt;-&gt; GPS connections are only made between the GPS radio and the 'Master' EDM. None are made to the 'Slave' EDM.</li> <li>b. If you do not have the fuel flow option, install the supplied P4 plug (JPI PN 700709-5) to the 'Slave' EDM.</li> </ul> | <ul> <li>The main differences for the twin installation (versus a single) are:</li> <li>1. An interconnect cable (JPI PN 700725) is installed between the two instruments, allowing them to cross-communicate. See 'Illustration A' below.</li> <li>2. The right engine EDM acts as a 'Slave' unit and certain connections must be made as follows:</li> </ul> | Section 18 - EDM-851 TWIN ENGINE ADDENDUM<br>The following provides instructions for install / operation of the<br>EDM-851 Twin Engine configuration. Each EDM-830 instrument is<br>installed to its respective engine as if it were a single. One EDM<br>operates as a 'Master', and the other operates as a 'Slave'. This allows<br>each instruments fuel computer values, such as total fuel remaining,<br>total fuel used, hours of endurance and so on, to be combined and<br>displayed on the Master. |
|--------------------------------|------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|-------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
|                                |                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          |                                                                                                                                                                                                                                                                                                                                                                |                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             |

The following are reminders regarding installation requirements that must be followed:
Each EDM unit shall be grounded directly to the engine that it is monitoring (nor the aircraft ground dus).
Pilot Programming Mode and Factory Limits Mode should be performed on each EDM unit to match them to the installation requirements. NOTE: these parameters are intentionally not synchronized between the units. YOU MUST PROGRAM BOTH EDM's.



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|                      | _                                                                                                                  |                                                                                                                                                                                                                                                                                               | -                                                                                                             |                                                                                                                                                                                                                                                                | >                                            | P                                                                                                                            |                                                                                                                                                              |                       |
|----------------------|--------------------------------------------------------------------------------------------------------------------|-----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|---------------------------------------------------------------------------------------------------------------|----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|----------------------------------------------|------------------------------------------------------------------------------------------------------------------------------|--------------------------------------------------------------------------------------------------------------------------------------------------------------|-----------------------|
| For Your Safe Flight | Reset Fuel Used to zero in flight 1. Tap STEP to find USD. 2. Hold both STEP and LF until the display shows .0 USD | <ol> <li>FUEL? N.</li> <li>Tap STEP to find RATE 4*.</li> <li>Tap LF to change from 1 to 9<br/>(this is the time, in seconds,<br/>between parameters during<br/>auto-scanning). Set to 0 to<br/>disable auto-scanning.</li> <li>Tap STEP to find END? Y,<br/>then tap STEP to exit</li> </ol> | Change Auto-Scan Indexing Rate<br>1. Hold both STEP and LF until<br>the display shows<br>PROGRAM, followed by | <ol> <li>Exclude a Parameter in Auto-Scan         <ol> <li>Tap STEP to select the<br/>parameter to exclude.</li> <li>Tap both STEP and LF.</li> <li>Decimal point appears before<br/>parameter name, indicating it<br/>is now excluded.</li> </ol> </li> </ol> | Automatic Scan<br>1. Tap LF.<br>2. Tap STEP. | Percentage View 1. Hold LF for three seconds. 2. 'NRW should disappear and EGT columns will return to absolute temp heights. | <ul> <li>analize View</li> <li>Hold LF for three seconds.</li> <li>'NRM' should appear and EGT,<br/>TIT &amp; CHT column heights<br/>will center.</li> </ul> | QUICK REFERENCE GUIDE |

 Transfer Data in Memory

 Tap STEP to clear any alarms.
 Insert USB Drive into the EDM data port.
 DUMP NEW should appear.
 Tap STEP to select the download choice (NEW, ALL or EVT).
 Tap STEP to initiate transfer.
 Wait for the normal engine monitor display to re-appear.
 Remove USB Drive.

 Initializing Tank Capacity 1. With power off, hold the STEP button in, then turn power on. Hold button in until the first item in table below 
 appears:

 MAIN-50
 in units selected

 AUX?
 Y-Yes-alioratic

 AUX=0
 Auxiliary tanks

 Auxiliary tanks
 Auxili

After this initialization, you should now add/subtract fuel in the EDM to make REM equal what is actually in your tanks.

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| Filled Tanks<br>Note: if 'f | ed Tanks<br>Note: if ' <i>FILL? N</i> ' is visible, start<br>('4.':        | Leaning<br>1. | Leaning Rich of Peak 1. Pre-lean mixture and wait one minute. 2 Tan I F (if you see I OP hold |
|-----------------------------|----------------------------------------------------------------------------|---------------|-----------------------------------------------------------------------------------------------|
| ,<br>                       | Hold both STEP and LF until the display shows PROGRAM, followed by ETH 2 M | ŗ.            | Tap LF (IX you see LOF, nom<br>both STEP and LF until you<br>see ROP).                        |
| n [r)                       | Tap LF to see FUEL? Y.                                                     | دىب           | Lean mixture until you see a column flash and the words                                       |
| ₽ÿ                          | See FILL? N. Tap LF to see                                                 |               | LEANEST followed by<br>1545 SET*                                                              |
| ςn                          | HLL /5"<br>With aux tanks or tabs, tap LF                                  | <u>4</u>      | To see the peak EGT of the                                                                    |
| 6                           | again to see FILL 120*<br>Tap STEP to accept.                              |               | first cylinder to peak and tuel<br>flow :1560 PK*or 1560 13.5*,<br>hold I E                   |
| Added                       | Added or Removed Fuel<br>Note: if 'FILL? N' is visible, start              | ş             | Enrich mixture to set desired temperature.                                                    |
| a. (4.):<br>I. H            | '4.':<br>Hold both STEP and LF until                                       | Leanin        | Leaning Lean of Peak                                                                          |
| ~                           | the display shows PROGRAM, followed by FUEL? N.                            | ÷             | minute.                                                                                       |
| -ر<br>.بس                   | Tap LF to see FUEL? Y.<br>Tap STEP.                                        | 2             | Tap LF (if you see KOP, hold<br>both STEP and LF until you                                    |
| 4                           | See FILL? N. Tap LF 2 or 3                                                 | ω             | see LOP).<br>Lean mixture until you see                                                       |
| S                           | Tap STEP and see .0 GAL.                                                   | 4             | inverted columns.<br>Continue leaning until column                                            |
| ġ                           | to decrease the amount of fuel                                             |               | flashes and the word RICHEST                                                                  |
| 7.                          | displayed.<br>Tap STEP to accept.                                          |               | of the last cylinder to peak and                                                              |
| Reset                       | Reset an Alarm                                                             | 5,            | To see the peak EGT of the                                                                    |
|                             | minutes): tap STEP.                                                        |               | delta fuel flow (GAMI spread),                                                                |
|                             | keset for remainder of logue,<br>hold STEP until the word OFF<br>appears.  | 6.            | Lean mixture to set desired temperature.                                                      |
|                             |                                                                            | *             | Values may vary depending                                                                     |
|                             |                                                                            |               | on your installation.                                                                         |
| $\sim$                      |                                                                            |               |                                                                                               |
| ę                           | Page 76                                                                    | Engine        | Engine Data Management                                                                        |
|                             |                                                                            |               |                                                                                               |
| ~                           |                                                                            |               |                                                                                               |

| Ort        |            | jeweils maximal                     | n der Leermas      |                                |            |             |         |                  |                  |                  |                  |                 |                 |         | Abzüge                   |             |           |           |           | Wägung       | Wägung un                                 |                            |                   |       | 1       | Nutzflugzeug (U) | (N) Fnerfinitian |          | LufttüchtigkGruppe |                                            |     | Bezugsebene   | Bezugspunkt       | Daten nach                             | Muster: Piper PA-28R-180 |              |
|------------|------------|-------------------------------------|--------------------|--------------------------------|------------|-------------|---------|------------------|------------------|------------------|------------------|-----------------|-----------------|---------|--------------------------|-------------|-----------|-----------|-----------|--------------|-------------------------------------------|----------------------------|-------------------|-------|---------|------------------|------------------|----------|--------------------|--------------------------------------------|-----|---------------|-------------------|----------------------------------------|--------------------------|--------------|
|            |            | joweils maximal zulässiger Füllung. | se sind enthalten: | (Dimension siehe Flughendbuch) |            |             |         | r lugebenalter 4 | riugelpenalter 3 | Flugelbehaiter 2 | Flügelbehälter 1 | Rumpfbehålter 2 | Rumpfbehälter 1 | Di      | Ausfliegbarer Kraftstoff | vorn/hinten | rechts    | links     |           | Auflage      | Wägung und Schwerpunktslage bei Leermasse |                            |                   | ł     |         | 8<br>C<br>       |                  | I        |                    | ,                                          | ? } | ₩.            | BP Flüg           | Daten nach Kennblatt bzw. Flughandbuch | PA-28R-180               |              |
| 15.04.2024 |            |                                     | !<br>:             | handbuch)                      |            |             |         |                  |                  |                  |                  |                 |                 | Dichte  | aftstoff                 | 62<br>      | G1 - 2    | G11       | Einheiten | Bru          | tslage bei L                              |                            |                   | -     |         |                  | 100,000          |          | lasse              | Antoniologi namoli i cisiki tri Makifettu. |     | 199 cm vor RP | Flügelvorderkante | v. Flughand                            |                          | -            |
| 4          |            |                                     |                    |                                |            |             |         |                  |                  |                  |                  |                 | -               | kg /l   |                          | 193,500     | 284,000   | 285,500   | ଜ୍        | Brutto-Masse | -eermasse                                 |                            |                   |       |         | 206,000          | 231,000          | X vom cm | Schwerpunt         |                                            |     |               | æ                 | ibuch                                  |                          |              |
| 1          | Leermasse  | Abzüge (Summe B)                    | Wägung (Summe A)   | Summe B                        | <br> -<br> | <br> -<br>T | <br>. - | <br> -<br>       | <br> -<br>T      | <br>- -<br>T     | <br> <br>T       | <br> -<br>T     | <br> -          | Einheit | Summe A                  | 0,000       | 0,500     | 0,500     | æ         | Tara-Masse   |                                           |                            |                   |       |         | 244,000          | 244,000          | -        | tslage bei Flug    | . waaijei eu t                             |     |               |                   | Grund d                                | Werk-Nr.: 30225          | Auftrags-Nr. |
|            | 762,000    | 0,000                               | 762,000            | 0,000                          |            |             |         |                  |                  |                  |                  |                 |                 |         | 762,000                  | 193,500     | 283,500   | 285,000   | ka        | Netto-Masse  | Plan der Flu                              | A                          |                   | -     | +-      | Kleiner 87       | 1135             | len kg   | bei Flugmasse      |                                            |     |               |                   | Grund der Wägung: Ä                    |                          | Ē            |
|            | 221,366    |                                     |                    |                                |            |             |         |                  |                  |                  |                  |                 |                 |         |                          | X2 52,000   | 279,000   | 279,000   | cm        | Hebelarm     | Plan der Fluggastraumgestaltung vom       | Ausrüstungsliste Stand vom | Ģ = Q = 20 = 24 = | <br>₽ | <br>    |                  |                  |          |                    |                                            |     |               |                   | Änderung Ausrüstung                    | Kennzeichen: D-EXRH      |              |
|            | 168673,500 | 0,000                               | 168673,500         | 0,000                          | 0,000      | 0,000       | 0,000   | 0,000            | 0,000            | 0,000            | 0,000            | 0,000           | 0,000           |         | 168673,500               | 10062,000   | 79096,500 | 79515,000 | kgcm      | Moment       |                                           | am 15.04.2024              | 0) - 1 - 07 - 11  |       | í,<br>S |                  | •                |          | 7                  |                                            |     |               |                   |                                        | 4                        |              |

| EDWF<br>Ort          | Einheil | Leermasse<br>762,000 kg             | Daten für den Eintrag ins Flughandbuch                    |                        |            |             | Höchstzulässige Zuladung | Gewichtstrimmung Einbauort | Einsatzausrüstung | Gepack |  |  | Sitzplätze: Flugzeugführer | Flügelbehälter 4 | Flügelbehälter 3 | Flügelbehälter 2 | Flügeibehälter 1 | Rumpfbehälter 2 | Rumpfbehälter 1 | Beladung   | Schwerpunktslage bei Flugmasse                                                                           | Einsalzausrüstung | Gepäck |    |    |   | Sitzplätze: Flugzeugführer | Flügelbehälter 4 | Flügelbehälter 3 | Flügelbehälter 2 | Flügelbshälter 1 | Rumpfbehälter 2 | (Rumpfbehälter 1 | Kraftstoff Dichte            | Veränderliche Lasten |
|----------------------|---------|-------------------------------------|-----------------------------------------------------------|------------------------|------------|-------------|--------------------------|----------------------------|-------------------|--------|--|--|----------------------------|------------------|------------------|------------------|------------------|-----------------|-----------------|------------|----------------------------------------------------------------------------------------------------------|-------------------|--------|----|----|---|----------------------------|------------------|------------------|------------------|------------------|-----------------|------------------|------------------------------|----------------------|
| 15.04.2024 A.SC みれどし |         | Leermasse-Moment<br>168673,500 kgcm | ns Flughandbuch Einheit                                   | höchstzul. Zuladung kg | - Leemasse | Höchstmasse | Luft                     | pr                         |                   |        |  |  | hrer                       |                  |                  |                  |                  |                 |                 | Leermasse  |                                                                                                          | E F               | ā      | Ka | Ka | æ | ×                          |                  |                  | _                |                  | _               | _                | te <b>rrest kg /</b> Einheit | 'n                   |
| ETC Stempel          |         |                                     | ι.                                                        | 373,000                | 762,000    | 1135,000    | Normalflugzeug (N)       | 762,000                    |                   |        |  |  |                            |                  |                  |                  |                  |                 |                 | 762,000    | Massen und Hebelarme sind dem Flughandbuch zu entnehmen<br>(mögliche vordere und hinkere Lage Xv und Xn) | 1,000             |        |    |    |   |                            |                  |                  |                  |                  |                 |                  |                              | Masse                |
| Ø                    |         |                                     | zusätzliche Angaben für Flughandbuch und Hinweisschilder: |                        |            |             | Nutzflugzeug (U)         | 221,356                    |                   |        |  |  |                            |                  |                  |                  |                  |                 |                 | 221,356    | belarme sìnd dem Flughand<br>/ und Xh)                                                                   |                   |        |    |    |   |                            |                  |                  |                  |                  |                 |                  | CTT                          | Hebelarm             |
|                      |         |                                     | sschilder:                                                |                        |            |             |                          | 168673,500                 |                   |        |  |  |                            |                  |                  |                  |                  |                 |                 | 168673,500 | dbuch zu entnehmen                                                                                       |                   |        |    |    |   |                            |                  |                  |                  |                  |                 |                  | kgcm                         | n j Moment           |